

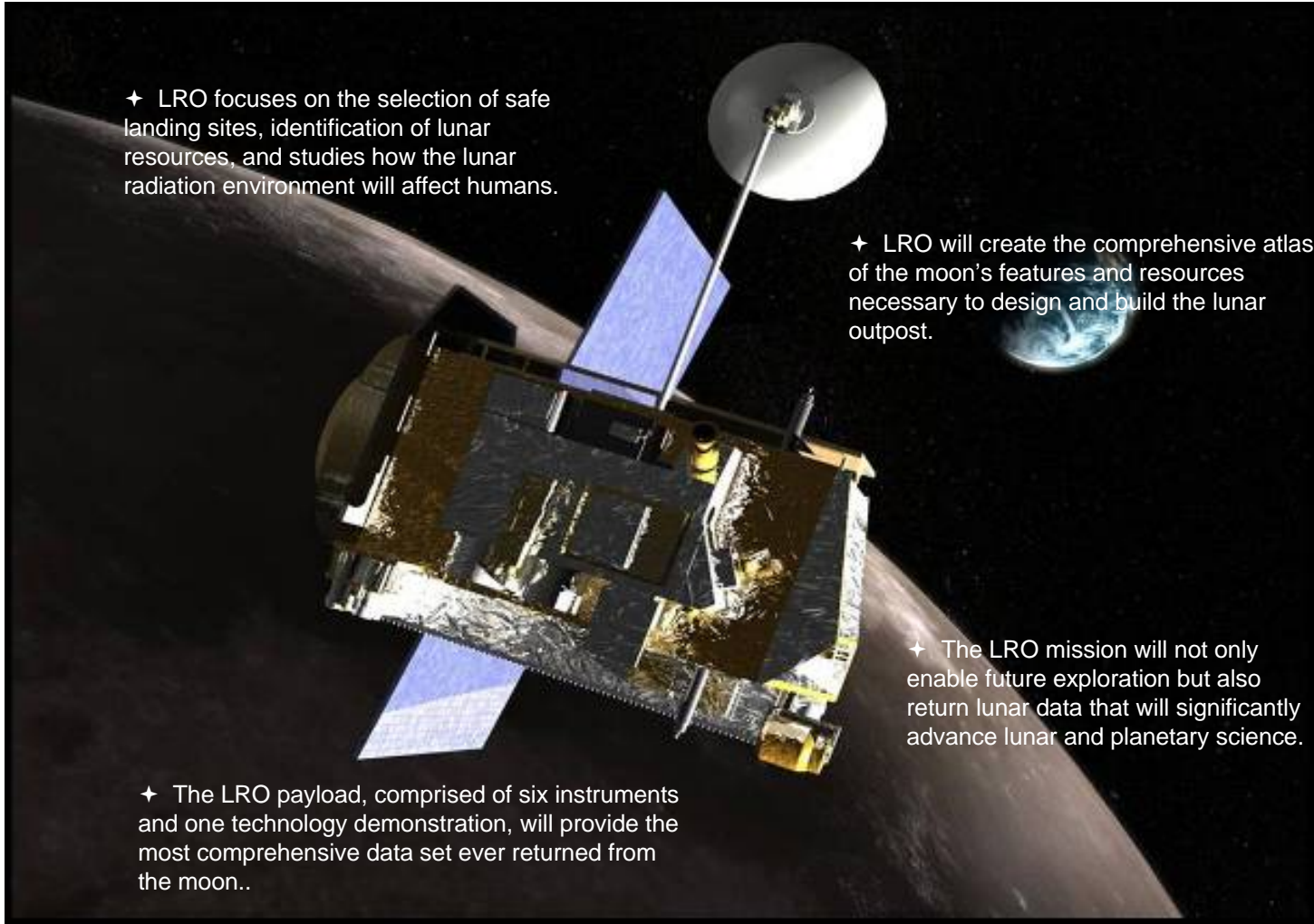


Lunar Reconnaissance Orbiter Overview





The Lunar Reconnaissance Orbiter (LRO) is NASA's first step in returning humans to the moon.



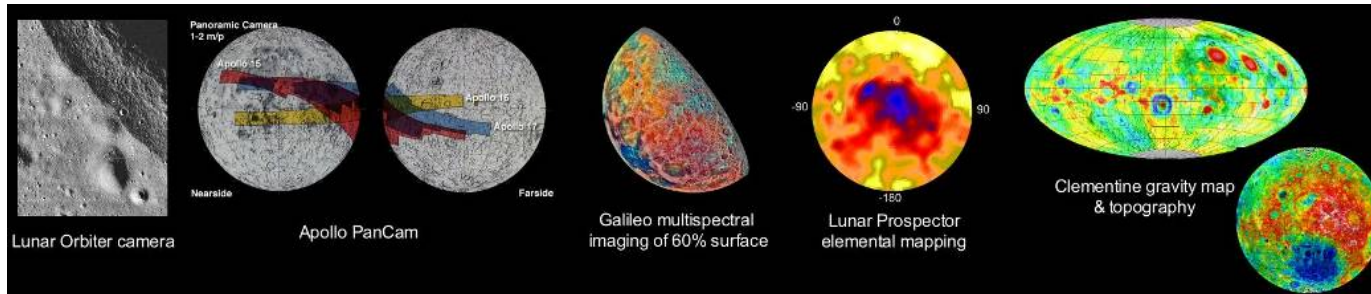
✦ LRO focuses on the selection of safe landing sites, identification of lunar resources, and studies how the lunar radiation environment will affect humans.

✦ LRO will create the comprehensive atlas of the moon's features and resources necessary to design and build the lunar outpost.

✦ The LRO mission will not only enable future exploration but also return lunar data that will significantly advance lunar and planetary science.

✦ The LRO payload, comprised of six instruments and one technology demonstration, will provide the most comprehensive data set ever returned from the moon..





Objective: The Lunar Reconnaissance Orbiter (LRO) mission objective is to conduct investigations that will be specifically targeted to prepare for and support future human exploration of the Moon.



Locate Potential Resources

Hydrogen/water at the lunar poles
Continuous solar energy
Mineralogy

Safe Landing Sites

High resolution imagery
Global geodetic grid
Topography
Rock abundances

Space Environment

Energetic particles
Neutrons



LRO Mission Overview



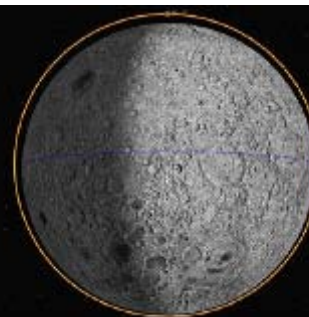
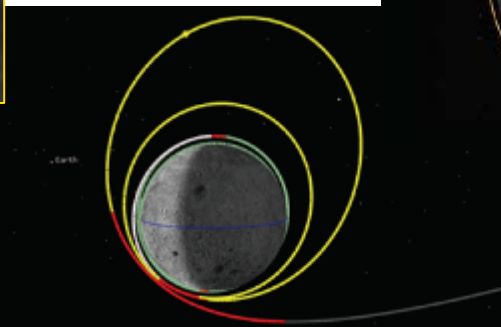
Launch: October 28, 2008



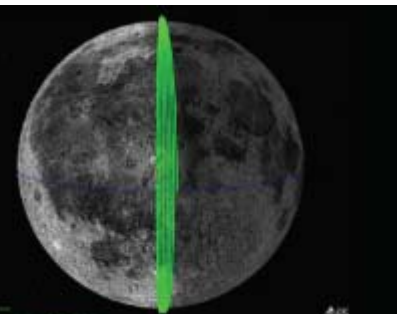
**Minimum Energy
Lunar Transfer ~ 4 Days**



**Lunar Orbit Insertion
Sequence, 4-6 Days**



**Commissioning Phase,
30 x 216 km Altitude
Quasi-Frozen Orbit,
Up to 60 Days**



**Polar Mapping Phase,
50 km Altitude Circular Orbit,
At least 1 Year**

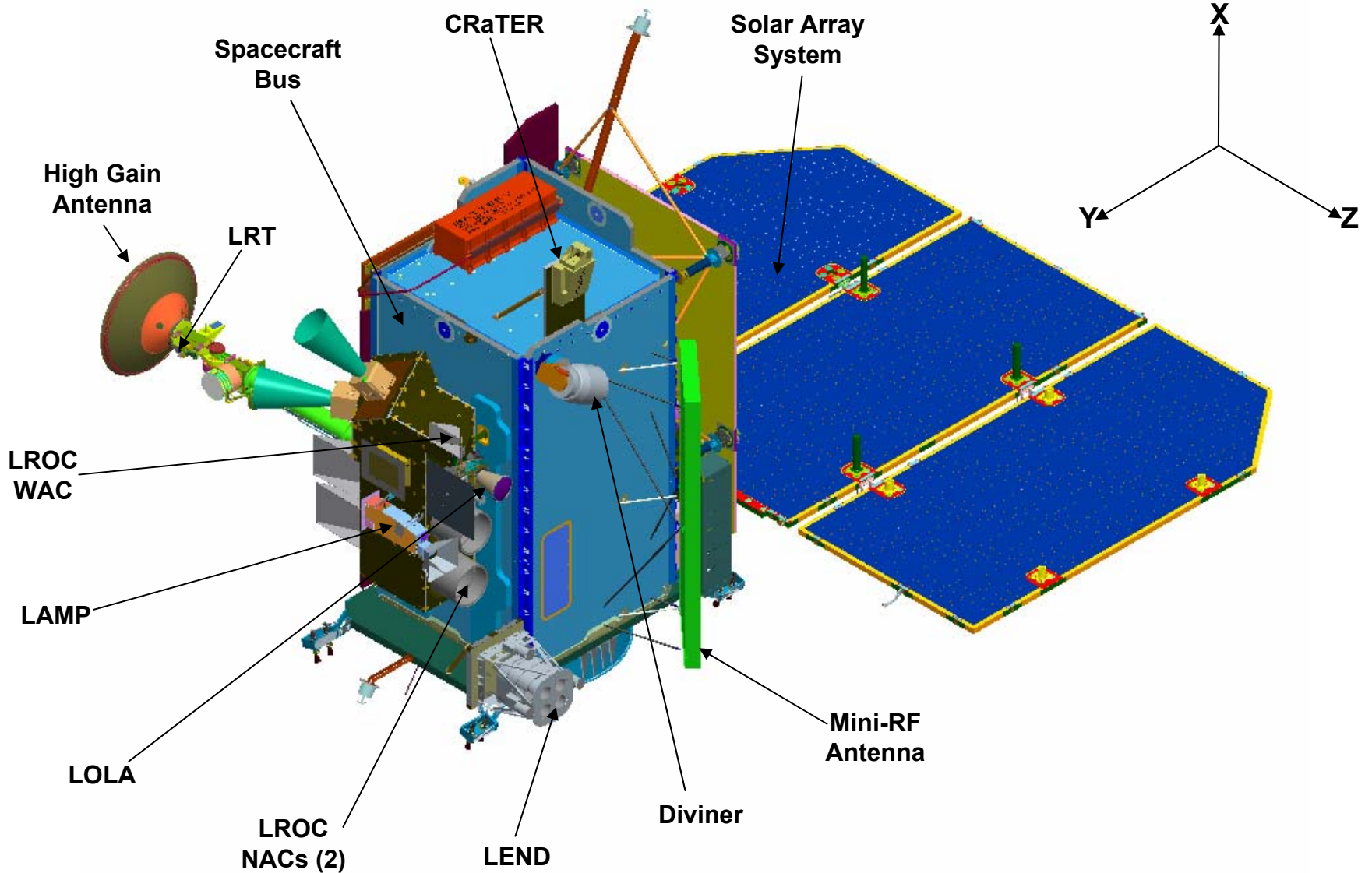


Nominal End of Mission: February 2010





LRO Spacecraft





Orbiter Instrumentation



LROC/WAC: Wide-Angle Camera

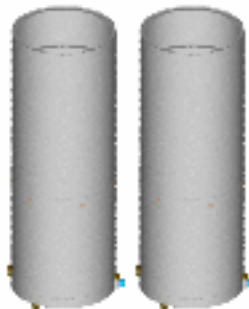
- Global Imagery
- Lighting
- Resources



Day Side
Autonomous

LROC/NACs: Narrow-Angle Cameras

- Targeted Imagery
- Hazards
- Topography



Day Side
Timeline Driven

LOLA: Lunar Orbiter Laser Altimeter

- Topography
- Slopes
- Roughness



Full Orbit
Autonomous

DLRE: Diviner Lunar Radiometer Exp.

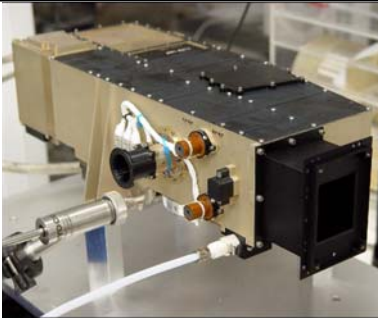
- Temperature
- Hazards
- Resources



Full Orbit
Autonomous

LAMP: Lyman-Alpha Mapping Project

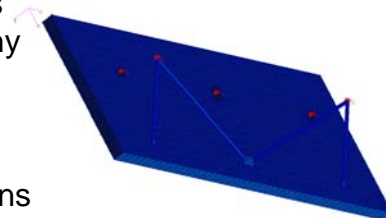
- Water-Frost
- PSR Maps



Night Side
Autonomous

Mini-RF: Synthetic Aperture Radar

- Tech Demonstration
- Resources
- Topography



Polar Regions
Timeline Driven

LEND: Lunar Explr. Neutron Detector

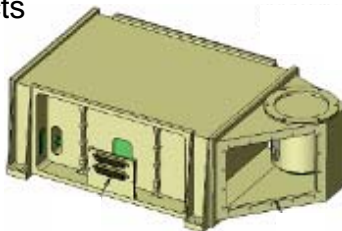
- Neutron Albedo
- Hydrogen Maps



Full Orbit
Autonomous

CRaTER: Cosmic Ray Telescope...

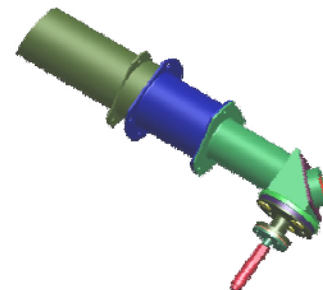
- Radiation Spectra
- Tissue Effects



Full Orbit
Autonomous

LRT: Laser Ranging Telescope

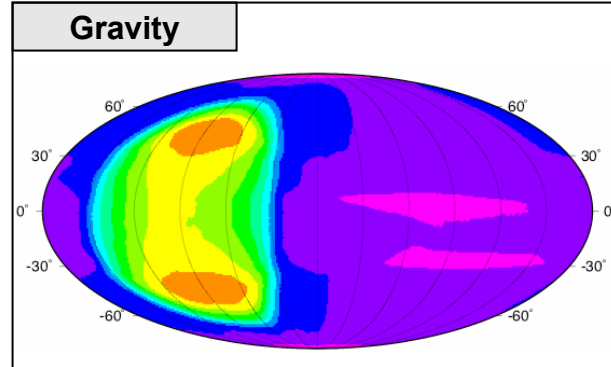
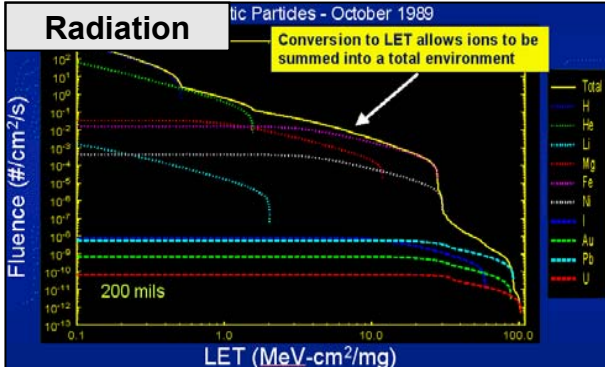
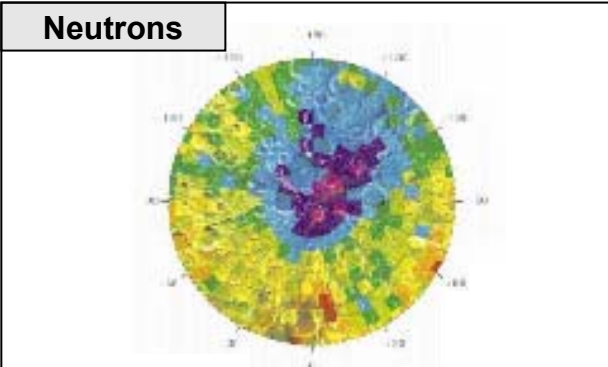
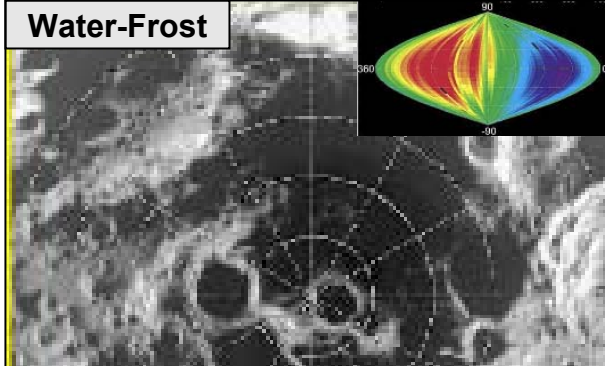
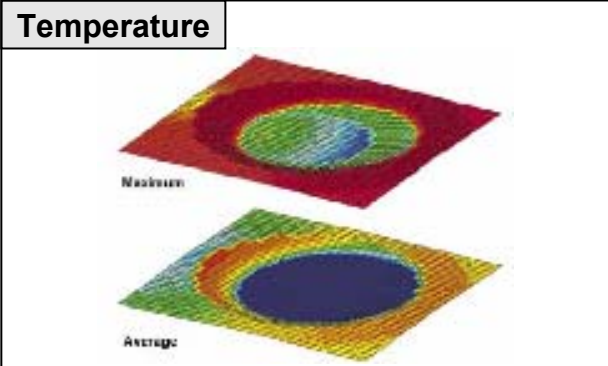
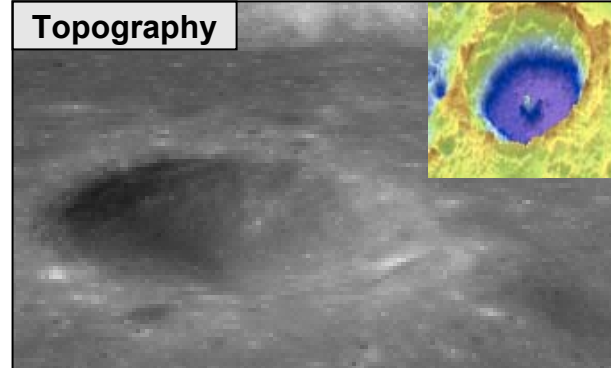
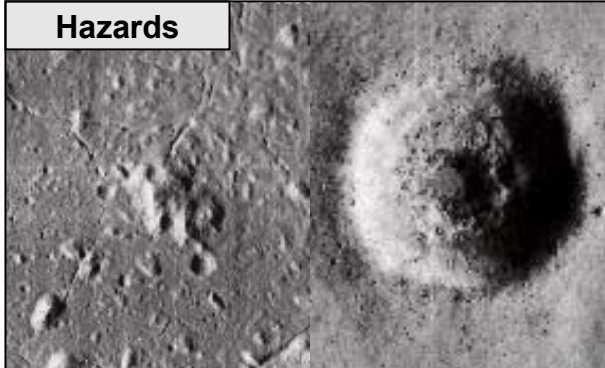
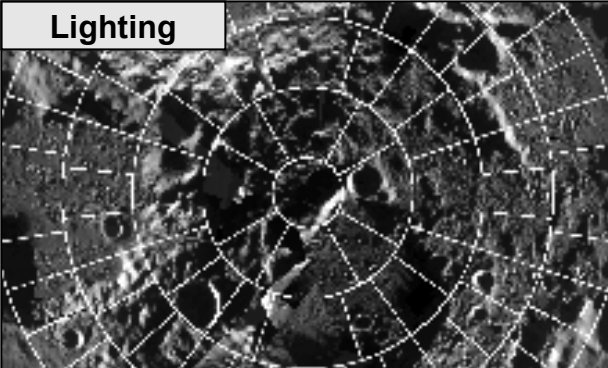
- Topography
- Gravity



LRGS LOS
Autonomous



Orbiter Data Products





LOLA Measurement Objectives



1. Topography of the Moon to an accuracy ± 1 meter and 0.1 meter precision.
2. Surface slopes in 2 directions to better than 0.5 degrees on a 50 meter scale.
3. Surface roughness to 0.3 meters.
4. Surface reflectance of the Moon at 1064 nm to $\sim 5\%$.
5. Establish a global lunar geodetic coordinate system.
6. Improve knowledge of the lunar gravity field

Along-track sampling in latitude 25 meters

Across-track sampling in longitude 0.04 degrees (~ 25 meters above lats 85 and ~ 1.2 km at the equator), after 1 year of operation.



LRO Spacecraft Positioning



- Simulations of the first 3 months of the LRO mission, and experience at Mars, suggest the addition of a precision range to the S-band tracking and inclusion of LOLA altimeter data can provide an improved model of the gravity field adequate for LRO orbit reconstruction.
- To improve the orbital position of LRO, and meet the LRO Level 1 Requirements for data products, improvement in the knowledge the lunar gravity field is needed.



Required Science Measurements by LR

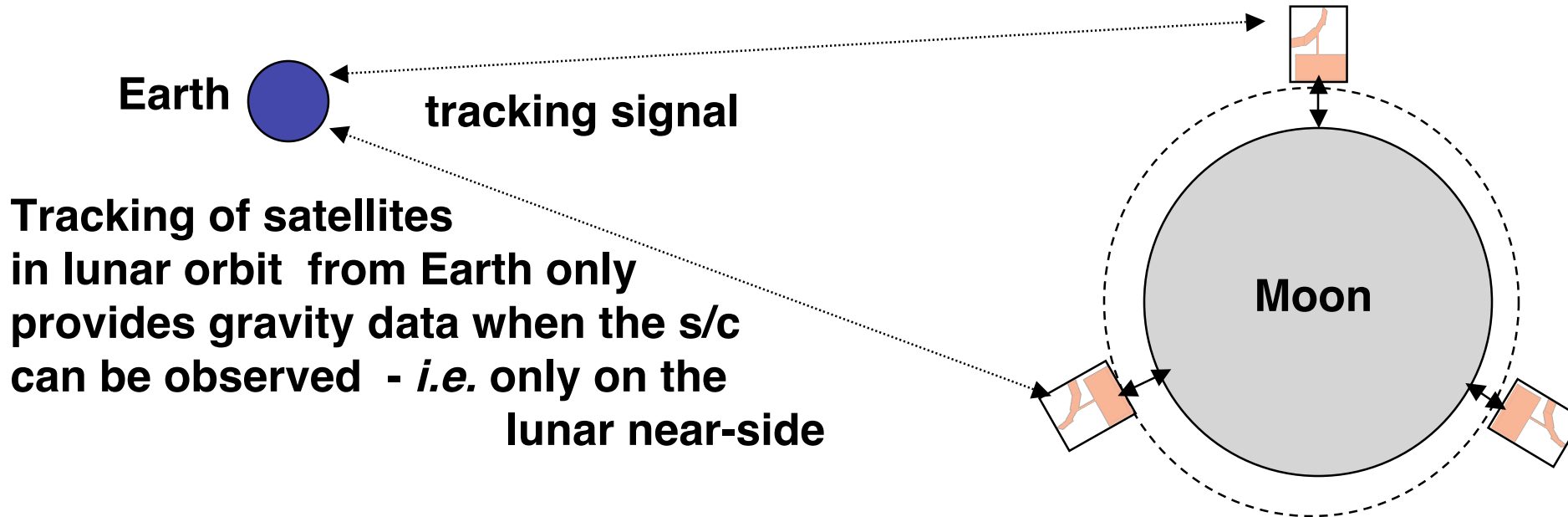


LR measurements will:

- Provide relative range measurements to LRO spacecraft at <10-cm precision, at 1 Hz.
- Maintain range stability to ± 1 m over 1 hour.
- in conjunction with the S-band data and the LOLA altimeter data, allow the orbit of the LRO spacecraft to be determined within 50 m along track, 50 m across track, and 1 m radial.
- improve the targeting of LROC by improving the prediction of the LRO orbit.
- allow every LROC pixel coordinate to be known to 50 meters and all LRO data co-registered at the 50-m horizontal level.
- improved knowledge of the lunar gravity model to enable visiting a particular location to within ~ 50 m.



Improving the Lunar Gravity Field



**Tracking of satellites
in lunar orbit from Earth only
provides gravity data when the s/c
can be observed - *i.e.* only on the
lunar near-side**

**Altimetry and altimetric cross-overs on the
far-side (and near-side) of the Moon will be
included as a tracking data for gravity
estimation**

**Cross-overs occur about every 1 to 2 km in
longitude and 3 deg in latitude at equator**



Laser Ranging





Acknowledgements



LOLA investigation Team:

David E. Smith, PI

Maria T. Zuber, Dep PI

Co-I's: Oded Aharonson (Caltech); James W. Head (Brown);
Frank G. Lemoine (GSFC) Gregory A. Neumann (GSFC/MIT);
Mark Robinson (Northwestern)

LOLA Instrument Scientist

Xiaoli Sun (GSFC)

LOLA and LR Instrument Design:

Jim Abshire, Xiaoli Sun, Jay Smith, John Cavanaugh, Luiz Ramos,
Danny Krebs, Jan McGarry, and many others