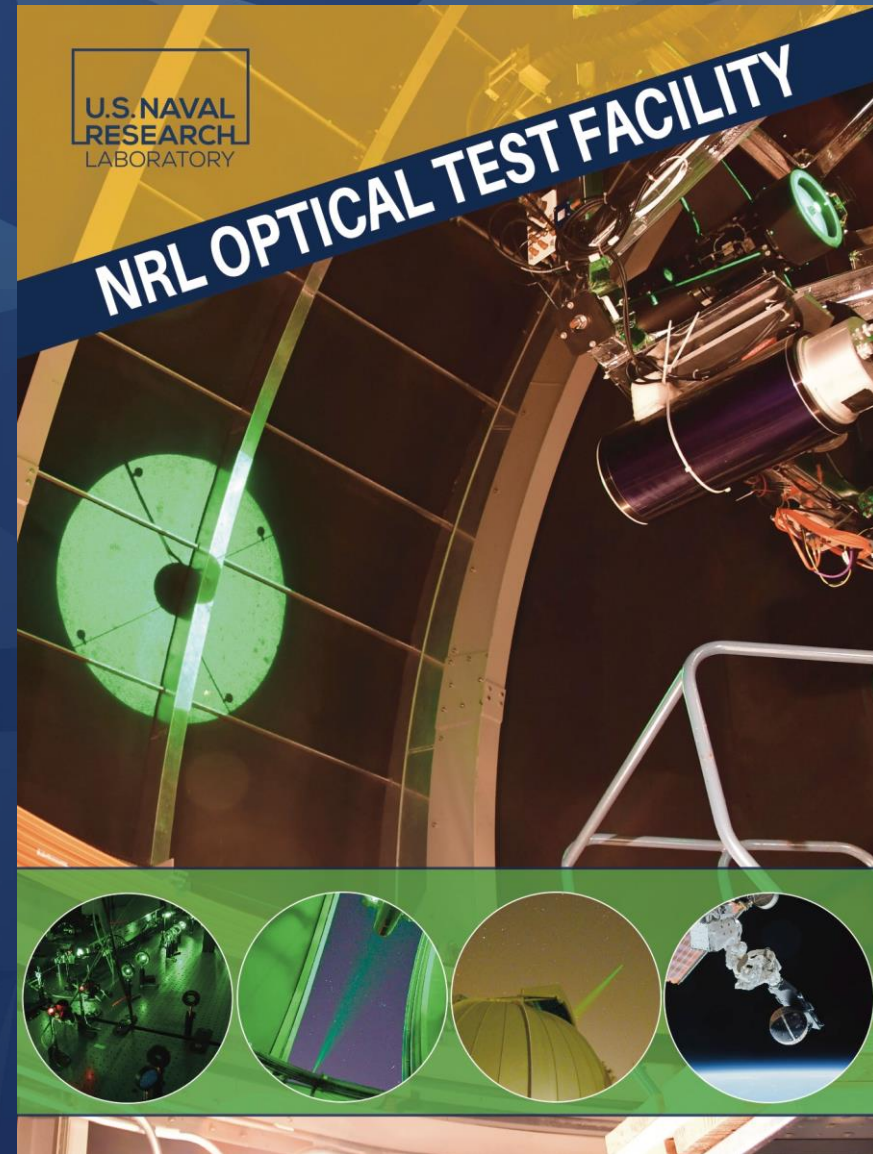




Status of Laser Time Transfer at Stafford, Virginia



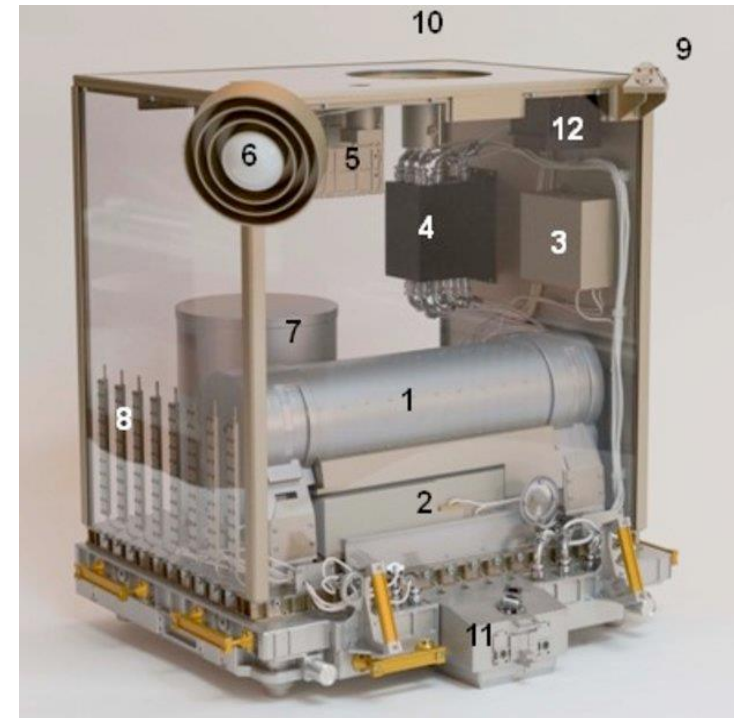
J. Griffiths, C. Font, R. Smith, A. DeRieux, L. Willstatter, C.I. Moore, F. Santiago, J. Ghorzi, L. Thomas

U.S. Naval Research Laboratory
Naval Center for Space Technology
Advanced Laser Technologies
Washington, DC USA

ILRS Technical Workshop – 21-25 October 2019 – Stuttgart, Germany

Objective & Motivation

- Enable NRL to participate in ILRS and other Laser Time Transfer (LTT) experiments, including CHOMPTT (NASA Ames/U. of FL) and ESA's ACES/ELT
 - leverage COTs equipment and technological advances from ILRS to date
- ACES/ELT
 - launch to ISS: expected in 2020
 - ultra-stable atomic (Cs fountain + H-maser) clock ensemble^{1,2,3}
 - microwave link¹⁰ for ACES primary time transfer mode
 - 532nm laser link¹² for optical timing experiments
 - gated detector: laser pulse on target within 100ns
- Objectives of optical link payload:
 - evaluate limits in comparing precision ground clocks via LTT utilizing ACES timescale
 - improve atmospheric propagation models by comparing refractive index to microwave propagation delay
 - optically derived precision orbits for ISS



Source: <http://www.esa.int>

- Goals since 21st IWLR
 - Close link to LEO satellite
 - Demonstrate 1 mm ranging precision
 - Demonstrate 1 cm or better ranging accuracy to LEO satellite
 - Calibrate transmit delays to within 1mm
 - Begin developments to control laser fire time

**Calibration &
Validation**

LTT Testbed: Hardware Setup

Telescope: NRL's Brashear 1m telescope

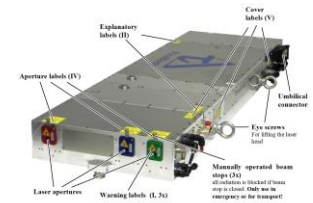
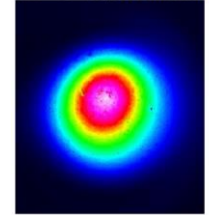
- All reflective design
- F89; Focal plane @12.640 m
- Slew rates:
 - El: 15 deg/s
 - Az: 25 deg/s
- Pointing accuracy: <2 arcsec RMS all sky



Laser: Lumentum PicoBlade

- Ultra-short pulses
 - ~34 ps (532 nm)
 - Single shot to 20 kHz
 - Approved for 5kHz ops
 - Operating @ 2kHz
 - 1.4 W avg power
- 82 MHz oscillator syncs to high precision external clock
- 1064nm capable

$\lambda = 532 \text{ nm}$
PRF = 1.5 kHz, 1000 mm from exit window



Detector: C-SPAD

- Si APD
- 200 μm active area
- Quantum Efficiency: 40%
- AR coated for 532nm
- Geiger-mode
- Time walk compensation <12ps



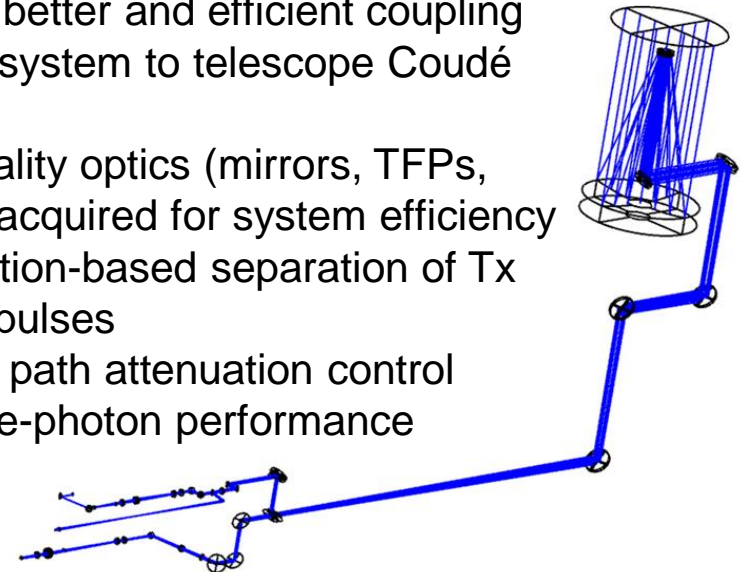
Event timers: NPET

- Supports 2+kHz epoch timestamping
- <0.9ps timing jitter per channel
- <0.5ps timing drift per Kelvin
- <0.1ps/hour timing stability
- Requires spectrally clean clock signal

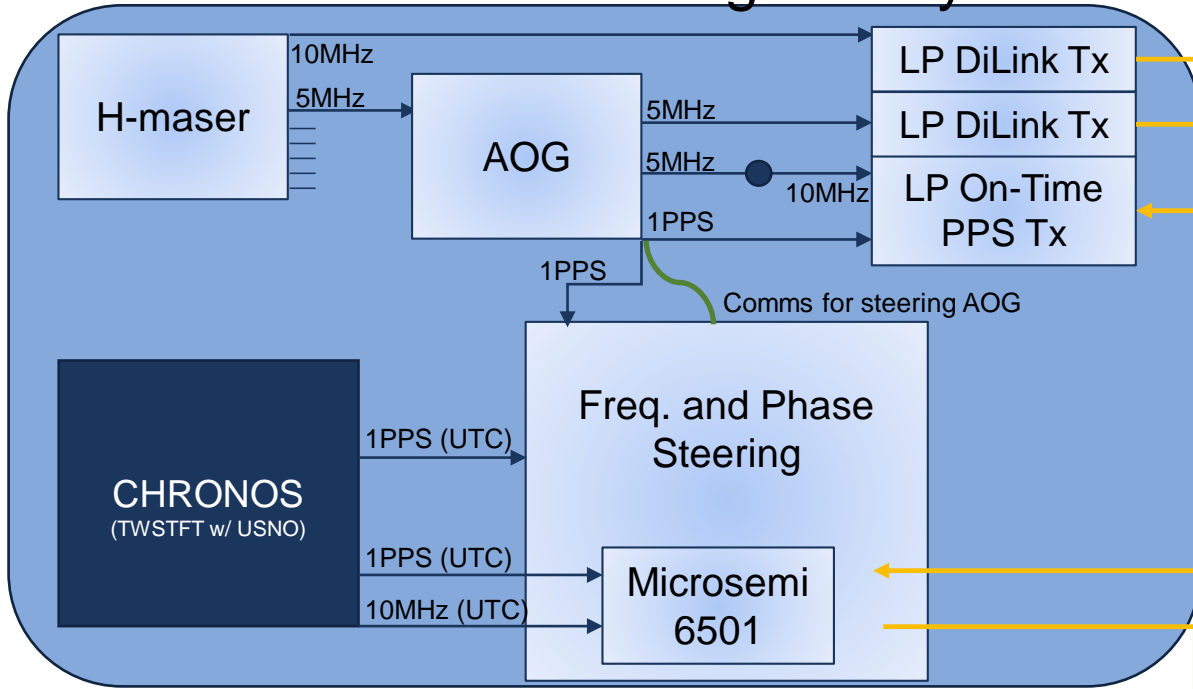


Optics:

- Custom optical elements designed at NRL for better and efficient coupling of laser system to telescope Coudé path
- High quality optics (mirrors, TFPs, lenses) acquired for system efficiency
- Polarization-based separation of Tx and Rx pulses
- Tx & Rx path attenuation control for single-photon performance



NRL CHRONOS Timing Facility



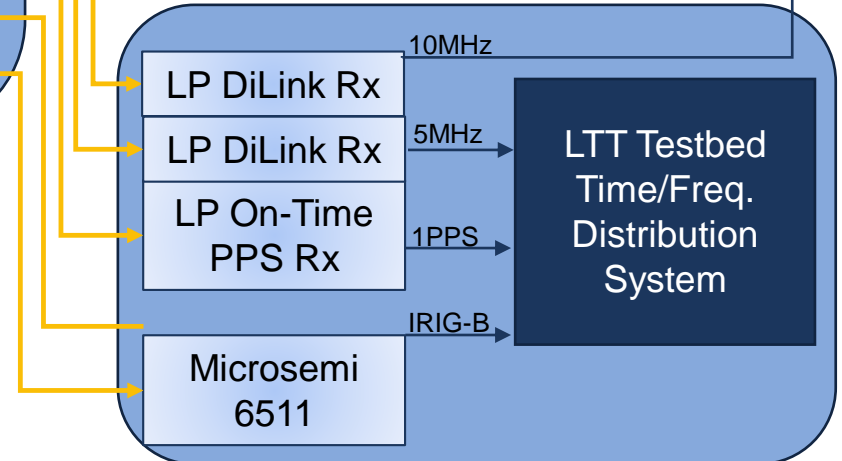
- 6511 IRIG used for time of day
- LP DiLink
 - uncompensated for dispersion effects and fiber path delay variations
- LP On-time PPS Tx – Rx offset: 276ps

MRC1 cGNSS



SMF-28e+
(~750m one-way)

NRL LTT Testbed



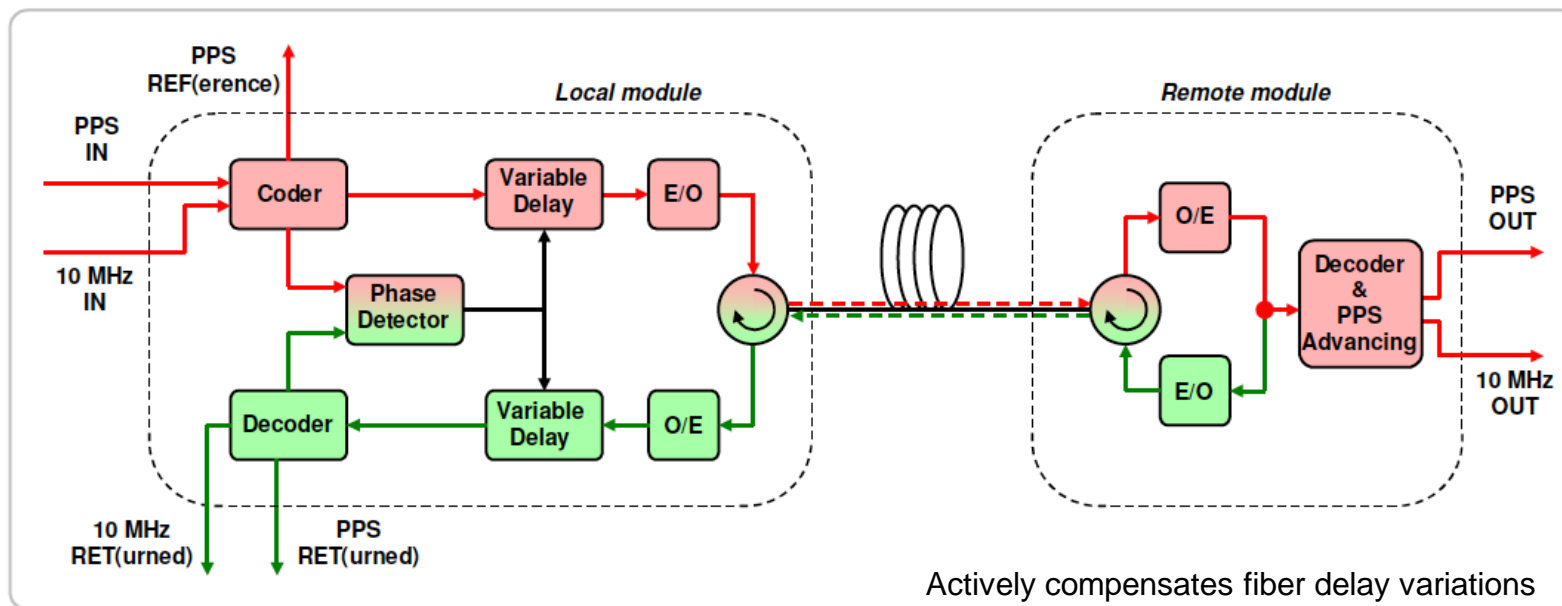
Upgrading LP units with PikTime OSTT-3 to improve H-maser+AOG distribution

PikTime OSTT-3 Fiber Time & Freq. Transfer

At Reference Clock



At LTT Testbed

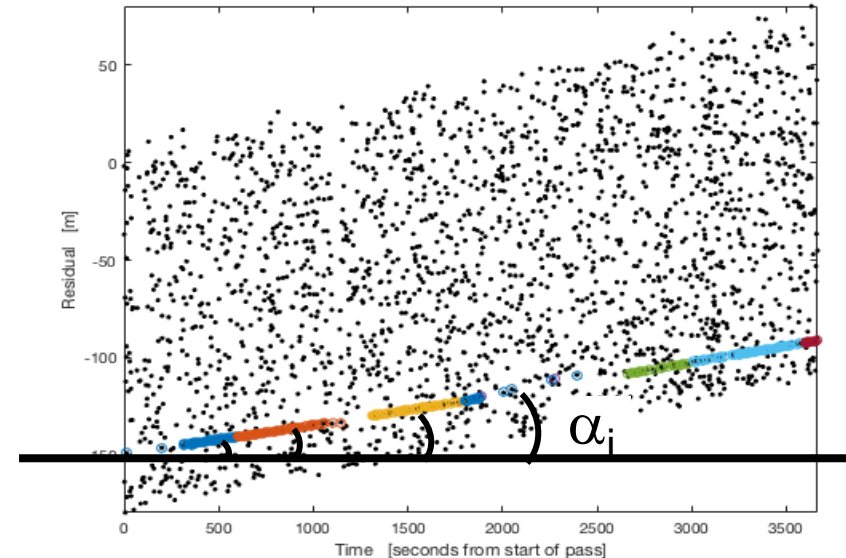


Time transfer stability: TDEV < 3 ps for 10 s averaging, < 1 ps for 10⁵ s averaging

Frequency transfer stability: ADEV < 3x10⁻¹³ for 1 s averaging, < 3x10⁻¹⁷ for 10⁵ s averaging

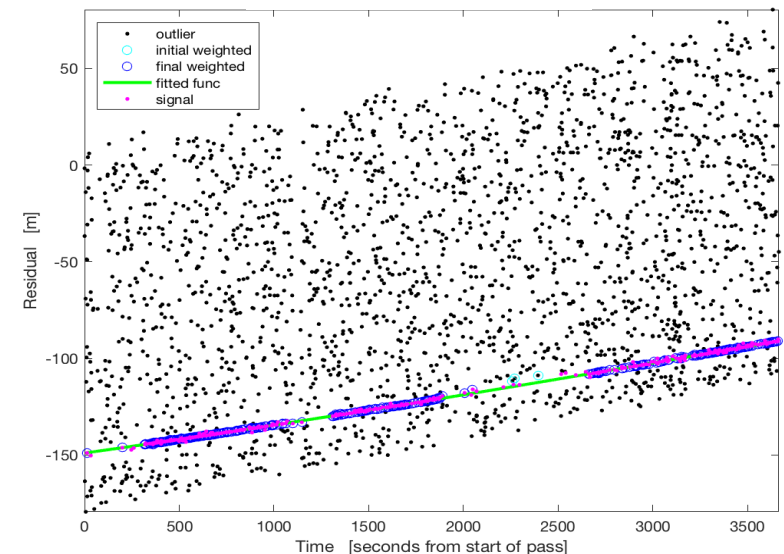
Step #1: Initial search

- Two-modes: ground calibration and satellite
- For each 10-sec interval, i ,
 - determine direction, α_i , that minimizes width of a high-resolution histogram and contains bin with maximal number of data points
 - select SLR measurements for residuals that fall within narrow band along direction α_i
 - thickness of the band is a function of system jitter and target signature



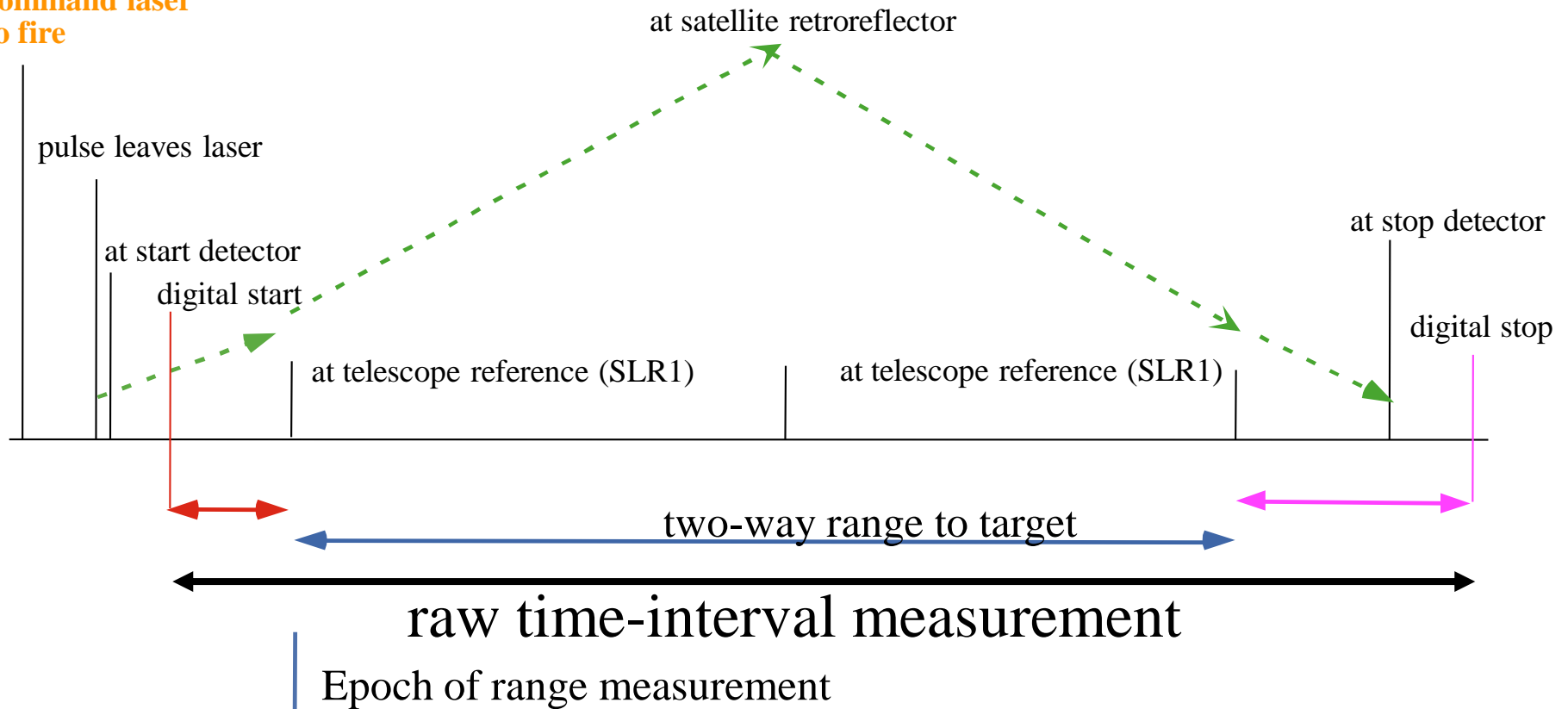
Step #2: Outlier rejection

- Iterative weighted least-squares of regression function to find signal photons
 - all data points are included
 - data found in Step 1 (cyan) used for initial weighting
 - subset of initial weighted data points remain (blue) after iterative fitting and outlier rejection
 - solution converges when no outliers remain
 - full-rate signal photons (magenta) are all remaining data points



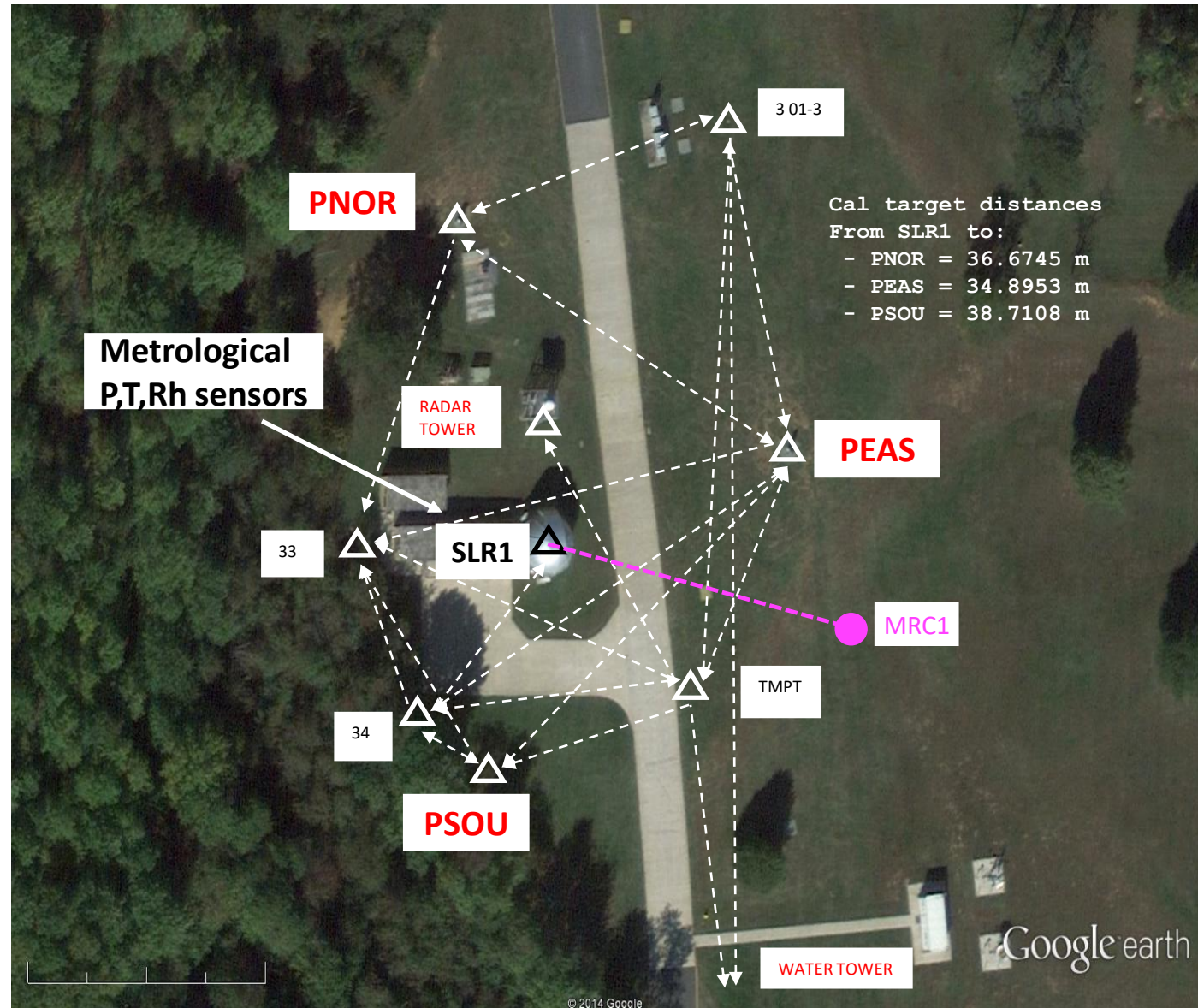
Calibration for LTT Experiments

command laser
to fire



- Determination of SLR1 and satellite position (state-of-art: ~30 mm 1D RSS error)
- Combined ranging delays [red + magenta] (GGOS goal: ~1 mm accuracy or better)
- Pulse transmission delay [red] (LTT goal: ~1 mm accuracy or better)
- Control laser fire [orange] to put pulses at SLR1 on-time (NRL goal: ~1 ns of UTC)

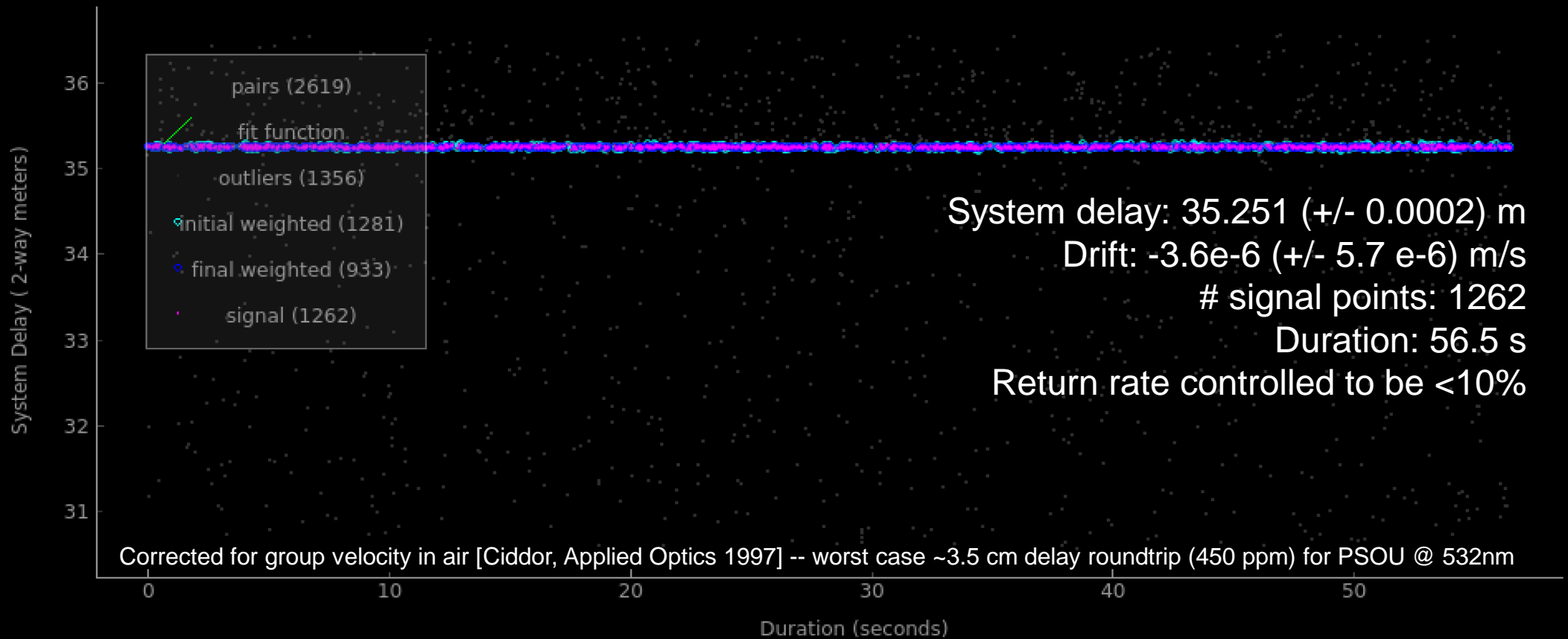
- Collaboration with NOAA National Geodetic Survey (NGS) Site Survey Team
- 2015-2016 Survey
 - developed techniques for realizing SLR1 and relative ground target distances
- July 2019 resurvey
 - new techniques for realizing SLR1
 - MRC1 – SLR1 tie
 - verified 2016 surveyed ranges to within ~0.5mm



LTT Testbed Range Calibration Precision

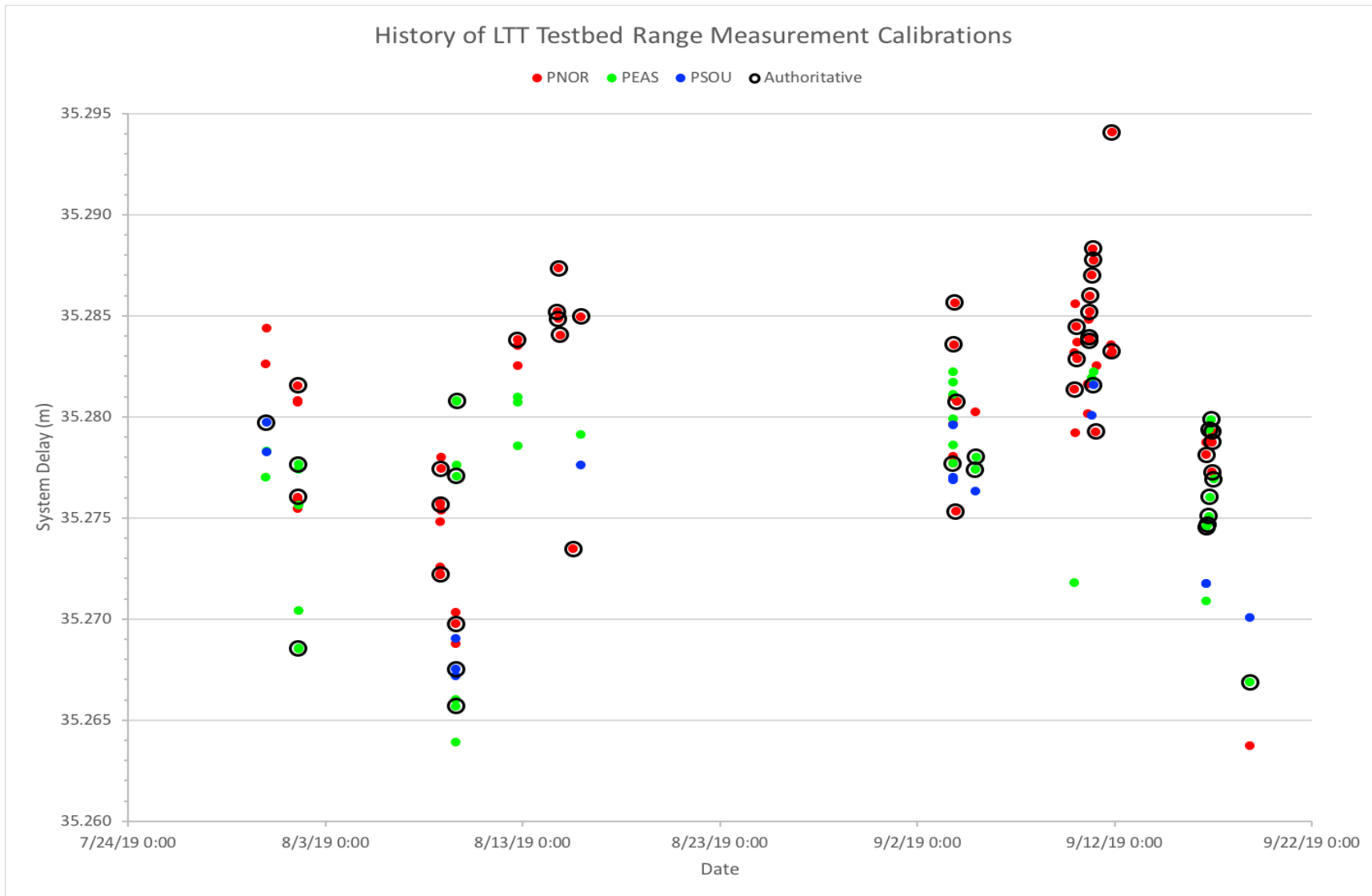
System Calibration

PEAS 2019-01-31 07:38 UTC



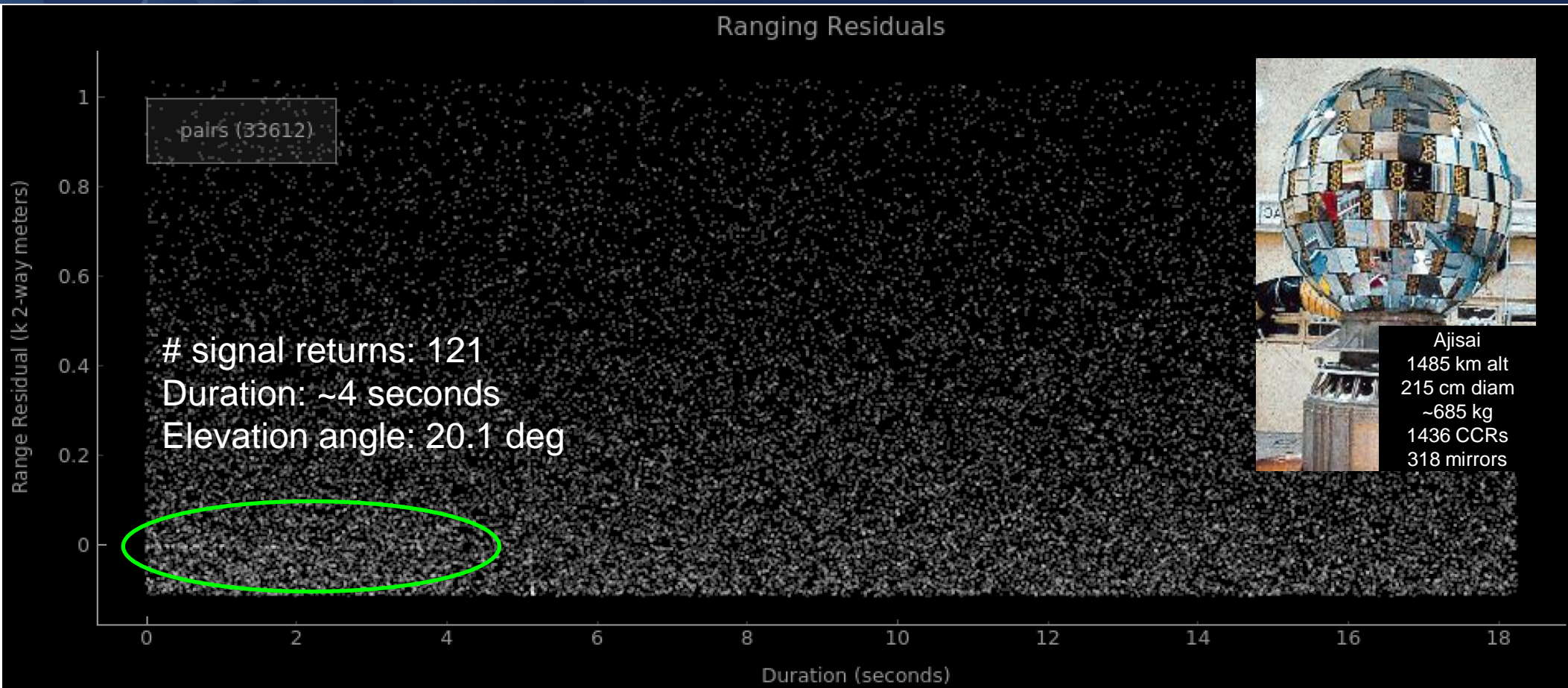
- Repeated collects on a ground cal target varies +/- 0.5 mm
 - Collect on PNOR (7 min later) diff from PEAS by 3 mm
 - PSOU collect (18 mins later) diff from PEAS by 1 mm
- Possible issue with CCR mount

LTT Testbed Range Calibration Stability



Back-to-back cals repeatable at ~3 mm level; drifts require cal <60 mins

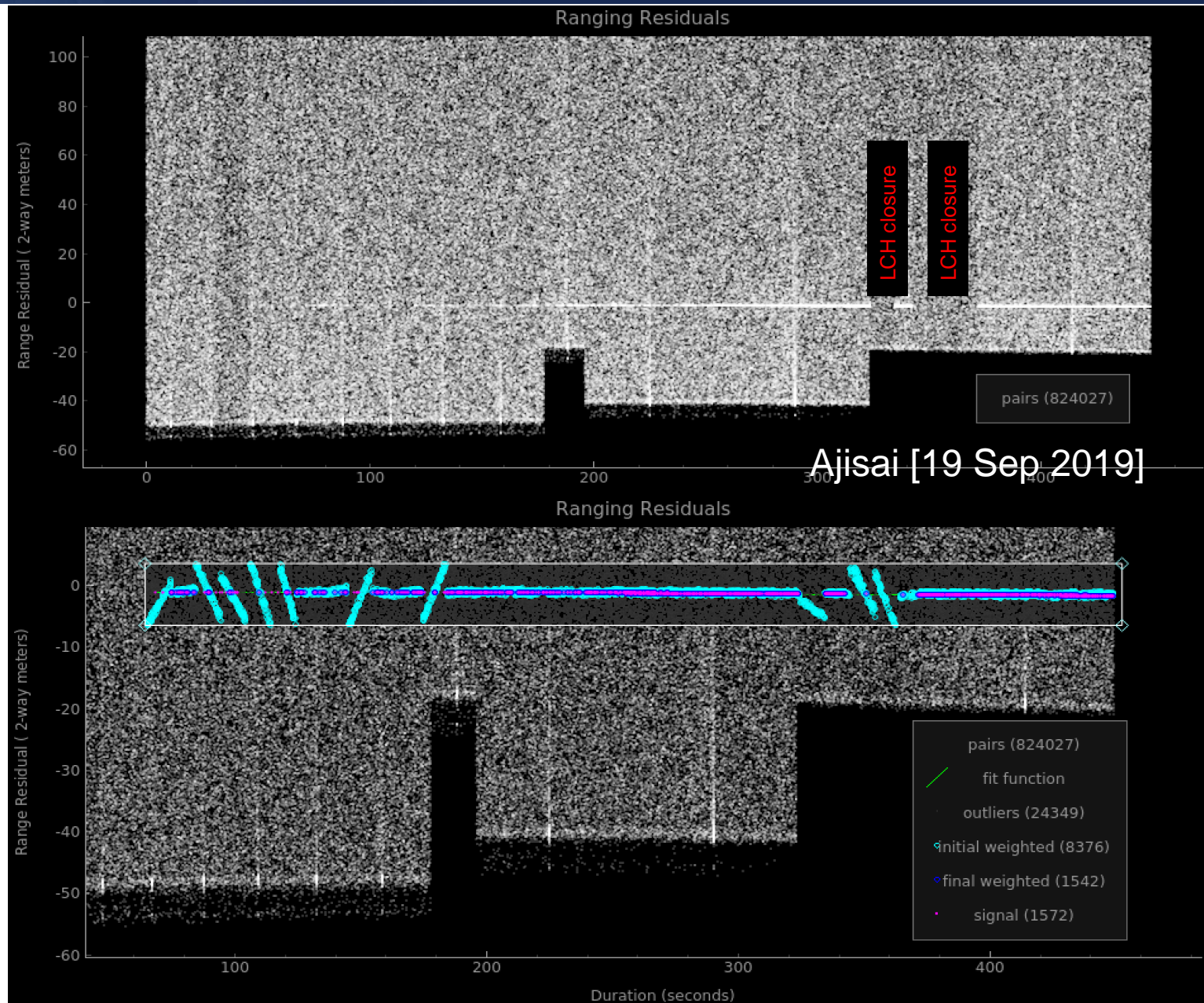
First light on 31 Jan 2019 (Ajisai)



Residuals near zero after correcting a 1s bias in event timer timescale
Additional Ajisai and Lageos collects in Sep 2019 after significant efforts to
refine and validate ground calibration performance

Analysis Filter for Extracting SLR Signal

- Generate all possible start/stop pairs for 2 kHz data
- Isolate and filter data
 - write signal data to Consolidated Ranging Data (CRD) product files used for precise OD



Validating LTT Testbed Range Accuracy

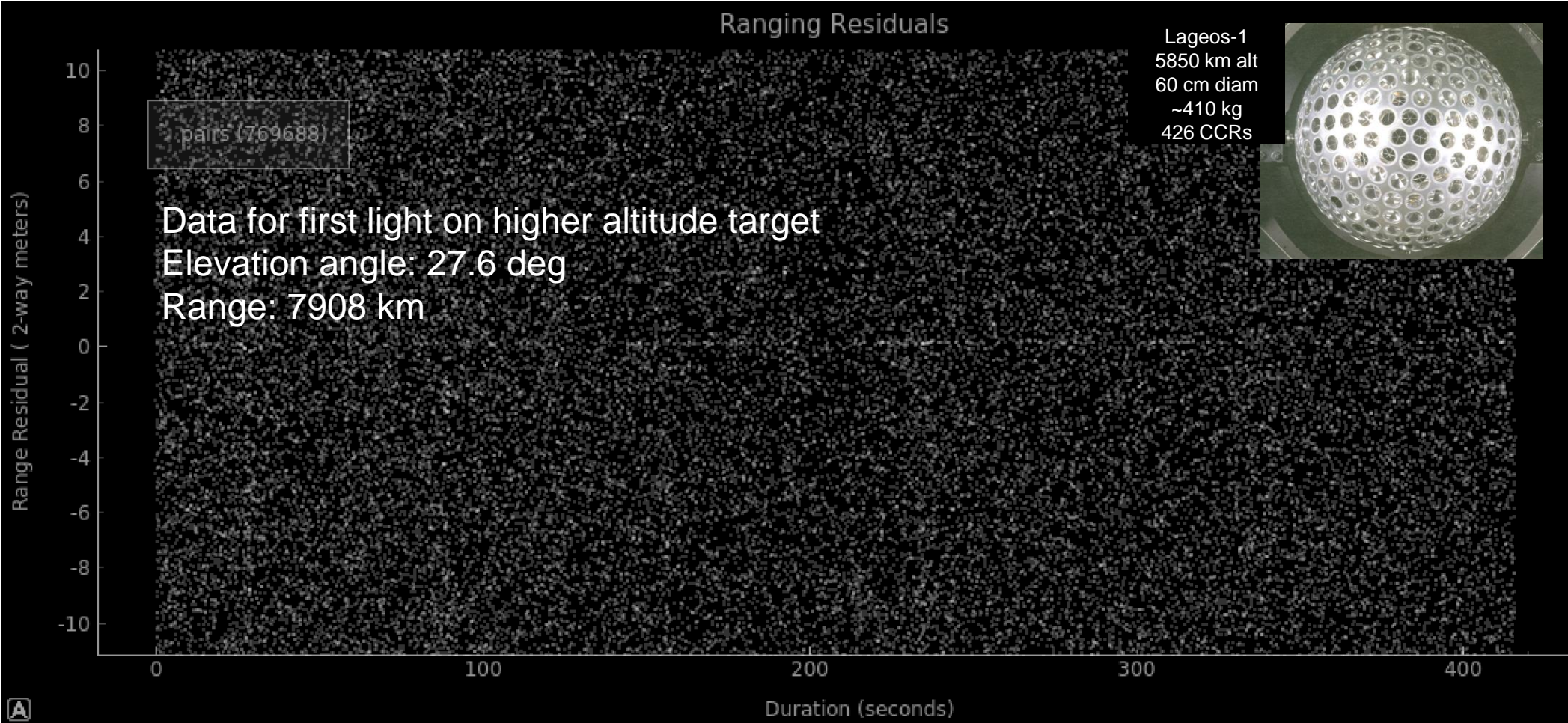
- Precision orbit determination (POD) using NRL and ILRS data for geodetic LEO satellites -- *Ajisai*, *Lageos*, *Starlette*, *Lares*, *Larets*
- New analysis tools used to extract NRL signal returns, then data used in precise orbit solutions
 - 7d arc solutions for Jan and Sep 2019 data sets
 - NASA's GEODYN POD s/w with latest models and methods consistent with IERS 2010 Conventions (Petit and Luzum, 2010)
 - Station positions not adjusted
 - Solved for station range biases per 7d orbit arc in addition to orbit initial state
 - *Ajisai* is dense spherical satellite above atmosphere—highly dynamical orbit
 - SLR-derived orbits using full ILRS network consistently ~2 cm 1D RSS accuracy



Ajisai POD Results

[31 Jan 2019] 1s time bias removed, 70 cm range bias, 5 cm RMS of post-fit residuals

[10-11 Sep 2019] ~0 s time bias, 2 cm range bias, 5 cm post-fit RMS



Preliminary result based on in-house GEODYN POD analysis
[19 Sep 2019] ~0 s time bias, 1 mm range bias, 8 mm post-fit RMS

More recent L1 & L2 collections with much stronger signal returns

- Goals since 21st IWLR

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- Demonstrate 1 mm ranging precision
- Demonstrate 1 cm or better ranging accuracy to LEO satellite
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**Calibration &
Validation**

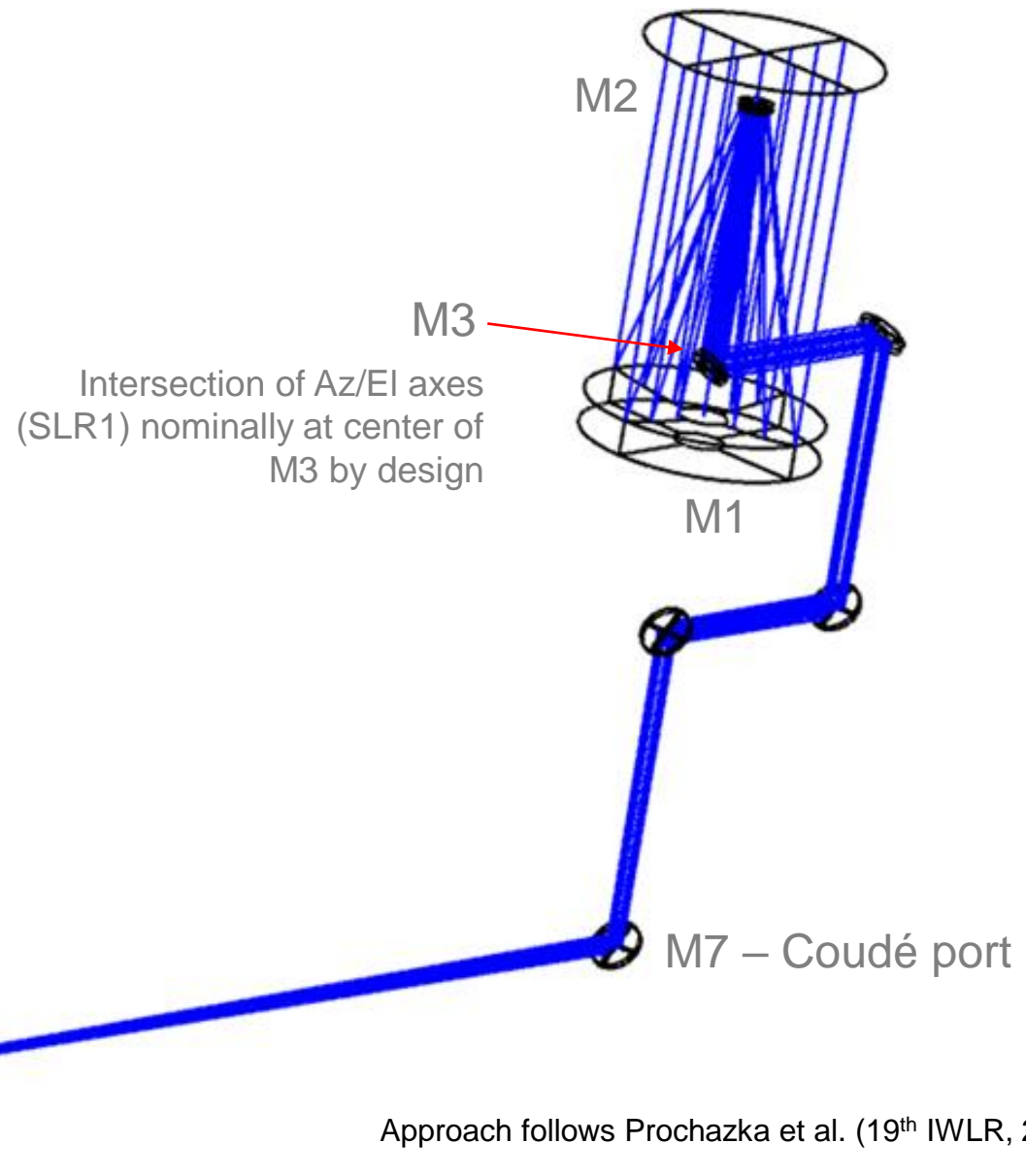
- Current Status

- Ground ranging repeatable to within 1 mm at each target, but up to 3 mm difference between targets (probably misalignment of target reflectors)
- First light to Ajisai (1485 km) 31 Jan 2019
- 2 cm ranging accuracy to Ajisai in Sep 2019, but results limited by satellite structure
- Potential <1 cm ranging accuracy (Lageos), but need more data
- Plan developed for calibrating transmit leg, waiting on hardware delivery
- Plan for controlling laser fire time, waiting on hardware delivery
- Currently working with NASA SLROC to submit quarantine data for ASC feedback

Questions?

• Plan

- Place detector at M2 or on spider
- Hang NPET on trunnion
- Provide 100MHz and 1PPS to NPET
- Time-tag start
- Time-tag arrival at M2
- Correct for M2-SLR1 path delay
- Monitor in real time
- Hardware on order



Control Laser Fire

H-maser LPN + AOG
(steered to UTC +/-1ns
via CHRONOS)

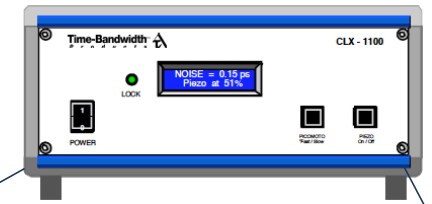
10MHz

SpectraDynamics LNFS-100
Low phase noise



82MHz
(+/- 2KHz)

CLX-1100
Laser Cavity Timing Stabilizer

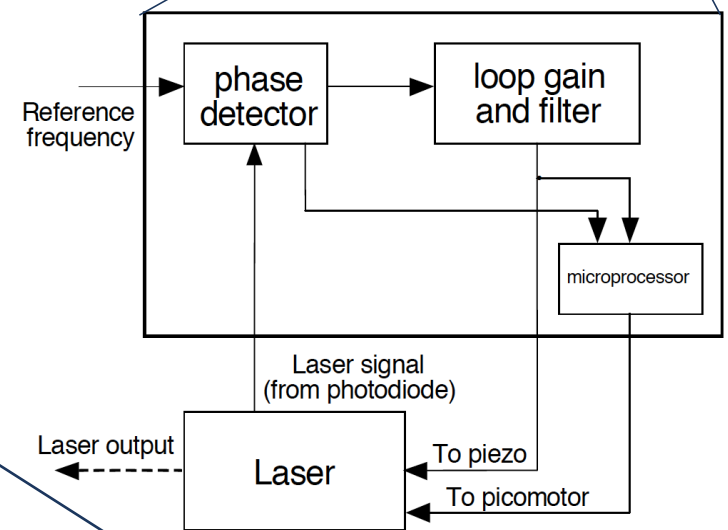


Software controlled frequency
steer +/- 2KHz for initial pulse
timing and maintaining rep
rate at satellite



Software generated trigger for
pick off

PoD for suppressing cw background
(for SLR @ 1064nm)



Trig w/ fixed delay